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Retardation and repair of fatigue cracks in a microcapsule toughened epoxy composite—Part II: In situ self-healing

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Abstract

Successful arrest and retardation of fatigue cracks is achieved with an in situ self-healing epoxy matrix composite that incorporates microencapsulated dicyclopentadiene (DCPD) healing agent and Grubbs' first generation Ru catalyst. Healing agent is released into the crack plane by the propagating crack, where it polymerizes to form a polymer wedge, generating a crack tip shielding mechanism. Due to the complex kinetics of healing a growing crack, the resulting in situ retardation and arrest of fatigue cracks exhibit a strong dependence on the applied range of cyclic stress intensity ΔK_I. Significant crack arrest and life-extension result when the in situ healing rate is faster than the crack growth rate. In loading cases where the crack grows too rapidly (maximum applied stress intensity factor is a significant percentage of the mode-I fracture toughness value), a carefully timed rest period can be used to prolong fatigue life up to 118%. At moderate ΔK_I, in situ healing extends fatigue life by as much as 213%. Further improvements in fatigue life-extension are achieved by employing a rest period, which leads to permanent arrest at this moderate ΔK_I. At lower values of applied stress intensity factor, self-healing yields complete arrest of fatigue cracks providing infinite fatigue life-extension.

Keywords: A. Smart materials; A. Polymer-matrix composites; B. Fatigue; D. Fractography, self-healing

1. Introduction

Self-healing materials are inspired by living systems in which damage triggers an autonomic healing response. White et al. [1] have developed a self-healing polymer that mimics many of the features of a biological system. Healing is accomplished by incorporating a microencapsulated healing agent and a catalytic chemical trigger within a polymer matrix. Damage in the form of a crack initiates the self-healing process, as does the fracture event in biological systems. The approaching crack ruptures the embedded microcapsules, releasing healing agent into the crack plane through capillary action. Polymerization of the healing agent is activated by contact with the embedded catalyst, bonding the crack faces.

Successful self-healing has been demonstrated for an epoxy composite modified with 5–25 wt% microencapsulated dicyclopentadiene (DCPD) monomer and 2.5 wt% Grubbs' first generation transition metal (Ru) catalyst. The embedded microcapsules were shown to rupture in the presence of a crack and release the DCPD monomer into the crack plane. Contact with the embedded Grubbs' catalyst [2] initiated ring opening metathesis polymerization (ROMP) of the DCPD [3,4] and rebonded the crack plane. This self-healing epoxy was

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able to recover over 90% of its virgin fracture toughness [5] and provide recovery from delamination damage in a reinforced composite [6,7]. In addition to providing an efficient mechanism for self-healing, the presence of DCPD-filled polymeric microcapsules also increased the inherent fracture toughness of the epoxy. Under monotonic loading the maximum toughness with microcapsules was 127% greater than neat epoxy [8]. The increased toughening associated with fluid-filled microcapsules was attributed to crack pinning along with increased hackle marking and subsurface microcracking. Brown et al. [9] also investigated the influence of microcapsules on fatigue crack propagation behavior of epoxy with the effects of self-healing precluded. The addition of microcapsules significantly decreased the fatigue crack-growth rate and increased the fatigue life above a transition value of the stress intensity factor [9].

In the current work, we investigate the performance of this successful self-healing epoxy system under cyclic loading (fatigue) conditions. Characterization of fatigue response is more complex than monotonic fracture due to dependence on the applied stress intensity range $\Delta K_I = K_{max} - K_{min}$, the loading frequency f, the ratio of applied stress intensity $R = K_{max}/K_{min}$, as well as the healing kinetics and any rest periods employed. Only a few studies of fatigue crack healing have been reported in the literature for polymeric materials. Daniel and Kim [10] investigated fatigue damage in asphalt by measuring the increase in the specimen compliance as microcrack growth occurred. After a rest period, gains in stiffness were observed and attributed to healing of the microcracks. Zako and Takano [11] performed a tensile fatigue test on a notched specimen to investigate crack healing in an epoxy composite. The specimen was fatigued until the stiffness decreased by 12.5%. The test was stopped and the crack was healed by application of heat, which triggered flow and subsequent polymerization of embedded particles of B-staged resin. The fatigue test was resumed with almost full recovery of stiffness. Following healing, the stiffness decreased at a similar rate to the virgin specimen. Both of these investigations considered successful healing as the recovery of stiffness lost due to damage induced by cyclic loading. Neither the effect on crack-growth rate or absolute fatigue life was considered.

In Part I of this paper [12], we reported successful healing of fatigue cracks through manual injection of precatalyzed DCPD resin into the crack plane. Healing efficiency was defined by the fatigue life-extension,

$$\lambda = \frac{N_{healed} - N_{control}}{N_{control}}, \quad (1)$$

where N_{healed} is the total number of cycles to failure for the self-healing sample and $N_{control}$ is the total number of cycles to failure for a similar sample without healing. Just after injection, the viscous polymer in the crack effectively shielded the crack tip and slowed crack growth. With time, a polyDCPD wedge was formed at the crack tip, leading to artificial crack closure and fatigue life-extension of $\lambda > 2000\%$. In Part II, we build on the success of these mechanisms for retarding fatigue crack growth to achieve the first demonstration of in situ self-healing of fatigue damage.

2. Fatigue test method

2.1. Materials and specimen preparation

Materials, specimen preparation and testing were nearly identical to that described in Part I of this paper [12]. Tapered double-cantilever beam specimens were cast from EPON® 828 epoxy resin (DGEBA) and 12 pph Ancamine® DETA (diethylenetriamine) curing agent with 20 wt% 180 μm diameter microcapsules [13] and 2.5 wt% Grubbs' catalyst mixed into the resin. The microcapsule concentration of 20 wt% was chosen to ensure adequate presence of healing agent in the crack plane. Control samples were also fabricated with no catalyst (only microcapsules) to preclude the effects of self-healing. The epoxy mixtures were degassed, poured into a closed silicone rubber mold and cured for 24 h at room temperature, followed by 24 h at 30 °C.

2.2. Mechanical testing

The fatigue crack propagation behavior of the self-healing epoxy was investigated using the tapered double-cantilever beam (TDCB) specimen geometry presented in Part I of this paper [12]. The TDCB geometry provides a crack length independent relationship between the applied stress intensity factor ΔK_I and load ΔP,

$$K_I = \alpha P, \quad (2)$$

where $\alpha = 11.2 \times 10^3\ m^{-3/2}$ for the current system [5]. Samples were precracked and immediately cyclically loaded. A triangular waveform of frequency 5 Hz was applied with a load ratio ($R = K_{min}/K_{max}$) of 0.1. Crack lengths were measured optically and by a calibration based on specimen compliance [9,12]. Each loading condition was investigated with continuous cyclic loading to sample failure and with rest periods to allow for healing with stationary crack faces. In all cases, fatigue crack growth in a self-healing sample was compared to that in a control sample (with no healing) under identical loading conditions and healing efficiency evaluated via Eq. (1).

3. Self-healing of the in situ system

In situ healing was investigated by measuring the fatigue life-extension of samples manufactured with

20 wt% microcapsules and 2.5 wt% catalyst. For successful in situ self-healing, the healing agent released into the crack plane must have enough time to polymerize. If the crack-growth rate is too fast, little or no healing will occur. In previous work, Brown et al. [5] measured the development of healing efficiency in this same materials system through monotonic fracture tests performed at prescribed times following the initial virgin fracture. After an initial dwell period of about 25 min during which no appreciable healing was measured, the healing efficiency increased rapidly, tapering off to a maximum healing efficiency after about 10 h (see Fig. 1). Comparison of the development of healing efficiency with the degree of cure (α) for bulk DCPD measured by Kessler and White [3] using differential scanning calorimetry (DSC) shows a similar exponential relationship with time (Fig. 1). Kessler and White [3] discussed the exponential functions to model the degree of cure are at length. Moreover, the development of measurable healing efficiency at 25 min closely corresponds to $\alpha = 1/3$, the theoretical gel point for a tetrafunctional monomer such as polyDCPD [14].

Anticipating that the competition between polymerization kinetics and mechanical crack growth would be a major factor influencing successful healing, three different levels of applied range of stress intensity ΔK_I were prescribed, one low-cycle fatigue case and two high-cycle fatigue cases. Low-cycle fatigue refers to the fatigue regime where ΔK_I approaches K_{IC} and rapid crack-growth causes sample failure after very few cycles (<10,000 cycles). High-cycle fatigue refers to the fatigue regime of low ΔK_I, relatively slow crack-growth rate and longer fatigue life (>10,000 cycles).

3.1. In situ low-cycle (high ΔK_I) fatigue healing

Under low-cycle fatigue conditions ($N_{healed} < 10{,}000$), crack propagation in the self-healing epoxy proceeded at a constant rate (Fig. 2) comparable to a control sample with no self-healing. Control sample data were virtually identical to the self-healing sample data indicating no retardation was taking place due to healing because crack propagation was so rapid. Sample fatigue life in the low-cycle fatigue regime was much shorter (2.5×10^4 cycles ~1.4 h) than the 10 h necessary for the healing agent to fully polymerize in the crack plane. The fatigue life-extension was essentially zero, $\lambda \sim 0\%$. The effect of rest periods was also investigated on two additional low-cycle fatigue cases. In the first case, loading was stopped after a small amount of crack growth and the samples were allowed to heal unloaded for 10 h to ensure full cure of the healing agent. The crack tip regressed to the approximate position of the TDCB notch, as shown in Fig. 3. However, after only a few cycles the crack tip rapidly progressed through the healed region to its location prior to healing. The fatigue healing efficiency λ for this case was essentially zero.

In the second case, fatigue loading was stopped after a small amount of crack growth and healing was allowed under load at K_{max} for 10 h. Fig. 2 shows the regression and retardation of a fatigue crack achieved in this case. Similar to the samples repaired by manual injection described in Part I of this paper [12], healing while loaded at K_{max} was much more effective. Under these conditions, polymerized healing agent formed a wedge at the crack tip, as shown in profile by optical microscopy in Fig. 4(a). Electron micrographs of the fracture plane (Fig. 4(b) and (c)), revealed that the polymer wedge consisted of a region of polyDCPD extending ~1 mm from the crack tip. The wedge

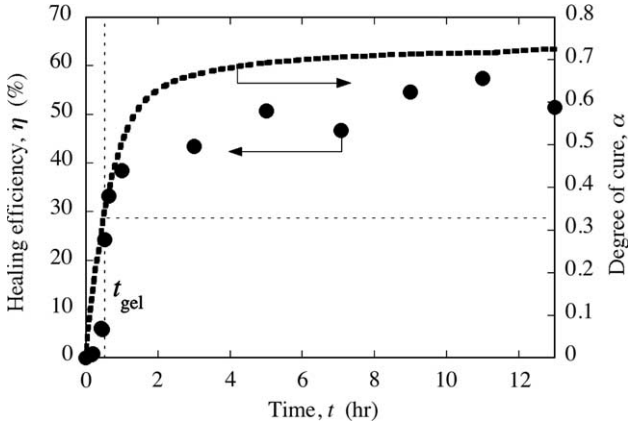

Fig. 1. Comparison between the development of healing efficiency and degree of cure of DCPD. Healing efficiency was obtained from monotonic fracture tests performed at prescribed times following the initial virgin fracture [5]. Differential scanning calorimetry (DSC) was used to measure degree of cure data for 30 °C isothermal polymerization of DCPD with 2 gl^{-1} of Grubbs' catalyst [3].

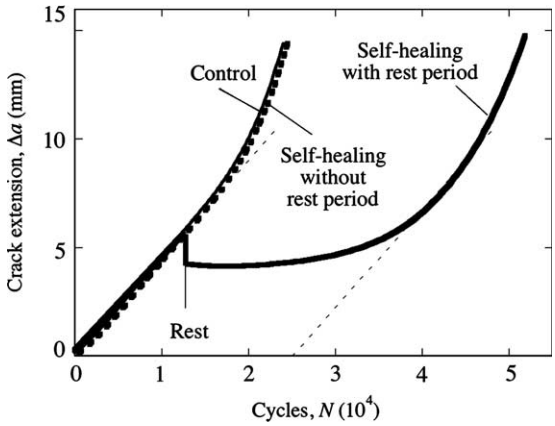

Fig. 2. Crack length vs. fatigue cycles of in situ sample tested in low-cycle fatigue regime without and with a rest period (under load at K_{max}), $\lambda = 0$ and 118%, respectively. $\Delta K_I = 0.405$ MPa m$^{1/2}$, $K_{max} = 0.450$ MPa m$^{1/2}$, $K_{min} = 0.045$ MPa m$^{1/2}$, $R = 0.1$, $f = 5$ Hz, and $a_0 = 31.2$ mm.

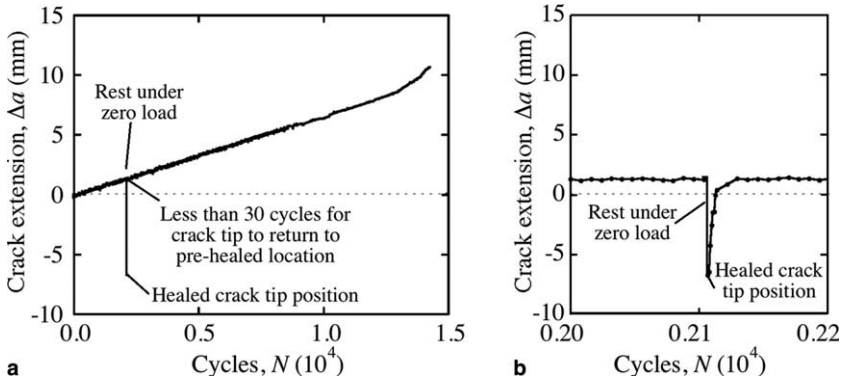

Fig. 3. Crack length vs. fatigue cycles of in situ sample with a rest period in the unloaded configuration, λ cong 0.2%. (a) Plotted for the entire fatigue life and (b) plotted in the region of the rest period. $\Delta K_I = 0.472$ MPa m$^{1/2}$ $K_{max} = 0.524$ MPa m$^{1/2}$, $K_{min} = 0.052$ MPa m$^{1/2}$, $R = 0.1$, $f = 5$ Hz, and $a_0 = 30.2$ mm.

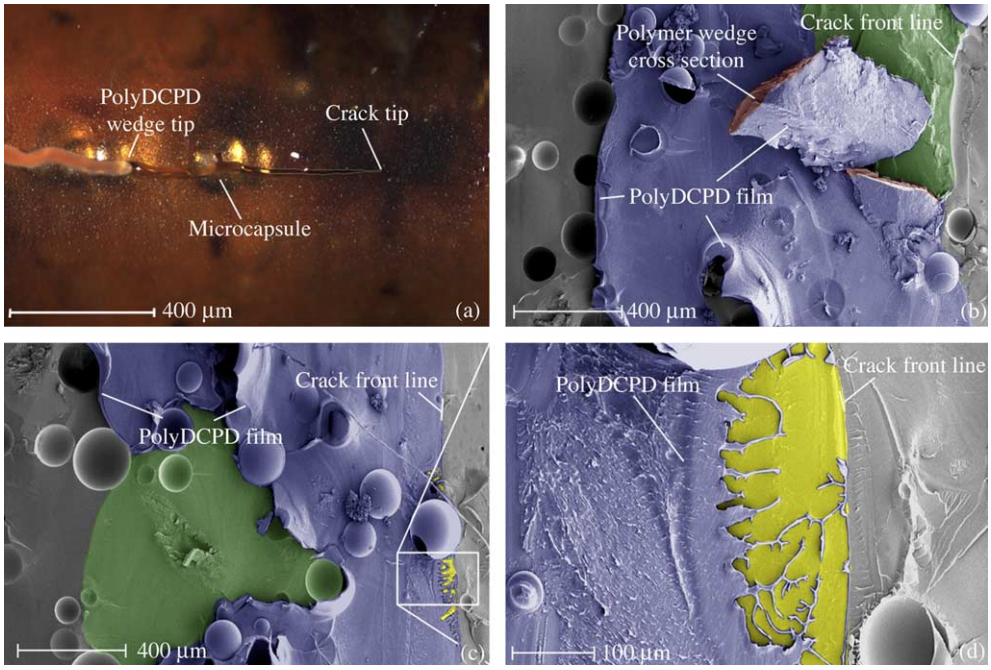

Fig. 4. PolyDCPD wedge at the crack tip of in situ sample tested in low-cycle fatigue regime with a rest period under load (see Fig. 2). (a) Optical micrograph of crack tip side view following rest period. The polymer wedge extends ~1 mm from the crack tip. (b) SEM micrograph of the fracture surface in the region of the crack tip. The polyDCPD wedge (blue) extends ~1 mm from the crack tip, with the cross section (red) tapering from a finite thickness to a sharp point at the crack tip. The region of the fracture surface where the polyDCPD has separated from the epoxy matrix is indicated in green. (c,d) PolyDCPD fills the crack to the tip along most of the crack front. In some regions, indicated in yellow, the DCPD does not penetrate fully to the crack tip. Note: The crack propagation is from left to right in all images.

penetrated into the sharp tip of the crack along the majority of the crack front line and had a significant out-of-plane thickness away from the crack tip. Because the interface was formed at K_{max}, it was under zero stress when the applied cyclic load reached K_{max} and under a compressive stress at all other points in the cycle. Under low-cycle fatigue conditions ($N_{healed} < 10{,}000$, high ΔK_I) fatigue life-extension λ for in situ self-healing epoxy with a rest period at K_{max} ranged from 73% to 118% for three samples. The 1–2 mm regression of the crack tip due to self-healing calculated from compliance measurements corresponded with direct microscopy measurements of the polymer wedge at the crack tip.

3.2. In situ high-cycle (low ΔK_I) fatigue healing

Under high-cycle fatigue conditions ($N_{healed} > 10{,}000$) the applied range of stress intensity ΔK_I was reduced, decreasing the crack-growth rate and increasing the number of cycles to sample failure. In this

regime, the sample fatigue life exceeded the time for the healing agent to gel (and quasistatic healing efficiency to develop). Self-healing fatigue life-extension was investigated for a number of samples under this type of loading. The effect of rest periods was also considered.

In situ samples were precracked and fatigued to failure. A typical plot of crack length vs. fatigue cycles is shown in Fig. 5. The initial release of healing agent during precracking retarded the crack growth, and led to some crack regression. Following this period of crack arrest, the crack eventually grew past the healed precrack ($\sim 3.5 \times 10^5$ cycles). After this point, the fatigue crack-growth behavior transitioned between periods of constant crack-growth rate and periods of crack retardation. During the periods of crack retardation, the crack-tip position corresponded to locations of exposed catalyst on the fracture plane. Polymerized healing agent was only present in the vicinity of exposed catalyst and formed an undulating structure with significant out-

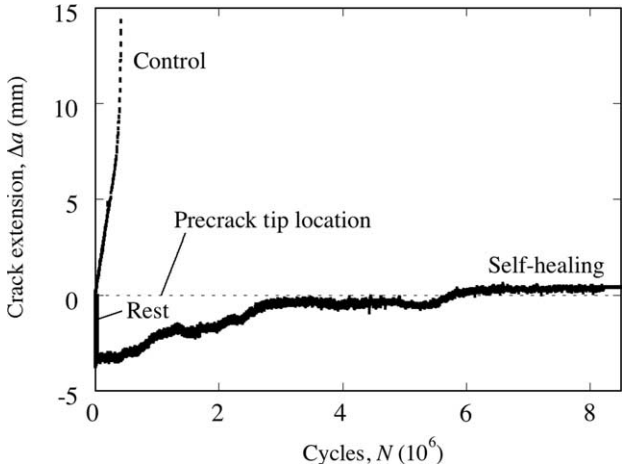

Fig. 7. Crack length vs. fatigue cycles of in situ sample tested in high-cycle fatigue regime with a rest period of 10 h at K_{max} after precracking, $\lambda = \infty$. $\Delta K_I = 0.338$ MPa m$^{1/2}$, $K_{max} = 0.376$ MPa m$^{1/2}$, $K_{min} = 0.038$ MPa m$^{1/2}$, $R = 0.1$, $f = 5$ Hz, and $a_0 = 31.6$ mm.

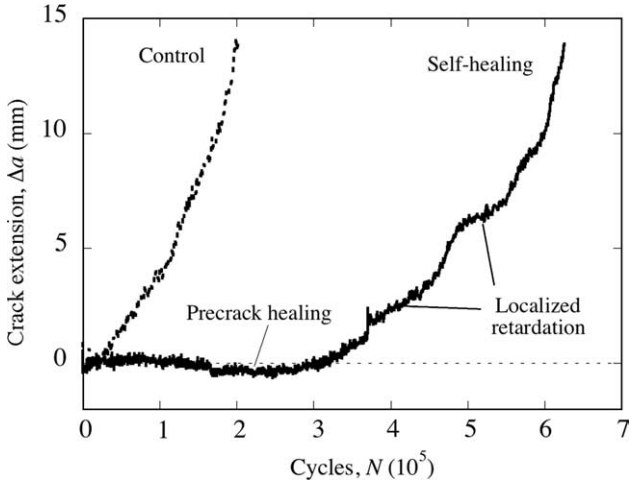

Fig. 5. Crack length vs. fatigue cycles of in situ sample tested to failure in high-cycle fatigue regime, $\lambda = 213\%$. $\Delta K_I = 0.338$ MPa m$^{1/2}$, $K_{max} = 0.376$ MPa m$^{1/2}$, $K_{min} = 0.038$ MPa m$^{1/2}$, $R = 0.1$, $f = 5$ Hz, and $a_0 = 35.4$ mm.

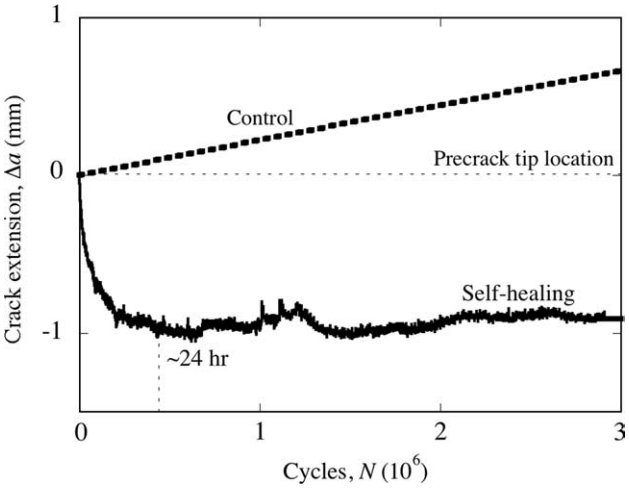

Fig. 8. Crack length vs. fatigue cycles of in situ sample in the threshold regime, $\lambda = \infty$. $\Delta K_I = 0.270$ MPa m$^{1/2}$, $K_{max} = 0.300$ MPa m$^{1/2}$, $K_{min} = 0.030$ MPa m$^{1/2}$, $R = 0.1$, $f = 5$ Hz, and $a_0 = 29.7$ mm.

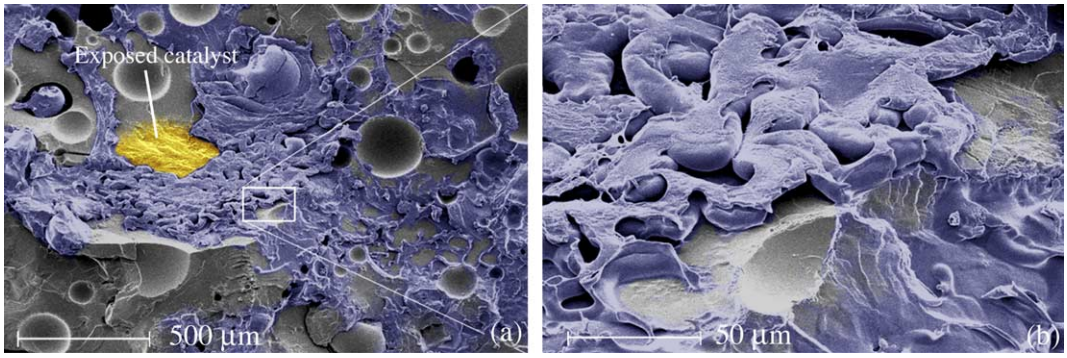

Fig. 6. (a) SEM micrograph of the in situ healed polyDCPD film (blue) on fatigue surface in the vicinity of a sites of exposed catalyst (orange). (b) The film forms an undulating structure due to cyclic loading during cure. Note: The crack propagation is from left to right in both images.

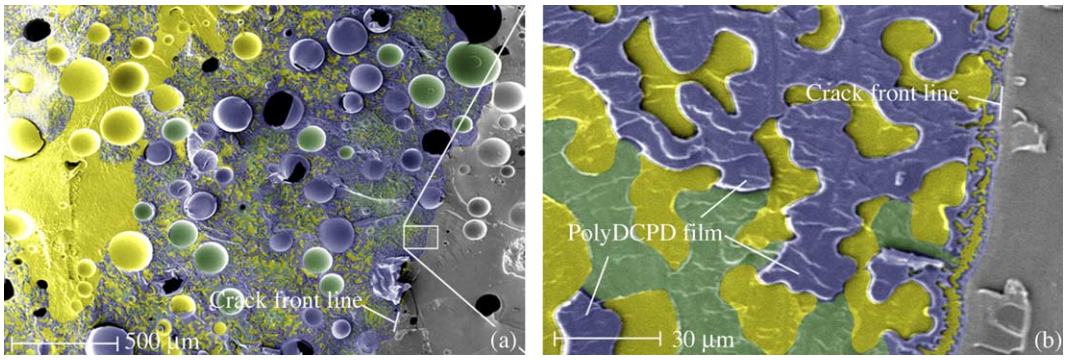

Fig. 9. SEM micrograph of the polyDCPD wedge at the crack tip of in situ sample tested in high-cycle fatigue regime resulting in crack arrest. (a) Image of the fracture surface in the region of the crack tip. The polyDCPD wedge (blue) extends from the crack tip forming a wavy pattern with many areas uncovered by polyDCPD (yellow). The region of the fracture surface where the polyDCPD has separated from the epoxy matrix is indicated in green. (b) PolyDCPD fills the crack to the tip along most of the crack front. Note: The crack propagation is from left to right in both images.

of-plane dimension, as evidenced by the fracture surface in Fig. 6. The local variation in concentration and imperfect dispersion of catalyst resulted in the localized periods of arrest. Total fatigue life-extension of six samples ranged from 89% to 213% and was amplified by increasing both the number of arrest events (i.e., increasing the number of catalyst particles exposed) and the duration of the individual arrest events. The recent development of wax-protected catalyst for self-healing by Rule et al. [15]—yielding improved dispersion of catalyst with increased reactivity—has the potential to provide more uniform an effective in situ healing in this loading regime, ultimately yielding even greater fatigue life-extension.

The fatigue life-extension due to precrack healing was dramatically improved by adding a rest period at K_{max}. In situ samples healed for 10 h at K_{max} following precracking and tested in the high-cycle fatigue regime exhibited permanent crack arrest in the two samples tested (see Fig. 7). As in the low-cycle fatigue case healed at K_{max}, a solid polyDCPD wedge formed at the crack tip during the rest period, with similar effect. The retardation elicited by the wedge was more efficient in this regime of loading. If K_{max} was reduced even further ($\Delta K_I < 0.5\ K_{IC}$), threshold conditions were achieved without a rest period. As shown in Fig. 8, the precrack regressed approximately 1 mm and never progressed further in the time frame of the test. In contrast, the precrack in the control sample was slowly growing at a constant rate. This effect was observed repeatably in four samples tested. Again, the healing agent released during the precrack event formed a partial polymer wedge at the crack tip (Fig. 9). Similar to the polyDCPD formed in the cycling crack under higher K_{max} (Fig. 6), the wedge created under lower cyclic load conditions had undulating surface features, but covered far more of the crack plane. A summary of life-extension values under the different loading conditions is given in Table 1.

Table 1
Fatigue life-extension from self-healing

Regime	Range of applied stress intensity, ΔK_I (MPa m$^{1/2}$)	Fatigue life-extension, λ	
		Continuously cycled to failure	With one rest period at K_{max}
$t_{fail} \ll t_{heal}$	0.7–0.9 K_{IC}	~0%	73–118%
$t_{fail} \sim t_{heal}$	0.5–0.7 K_{IC}	89–213%	∞[a]
$T_{fail} \gg t_{heal}$	<0.5 K_{IC}	∞	—

[a] Infinite fatigue life-extension denotes no optically measurable crack extension after at least 3×10^6 cycles (7 days of testing). Note: $t_{heal} \sim 10$ h from monotonic fracture [5].

4. Conclusions

The crack-growth behavior of self-healing epoxy under fatigue loading was investigated using a protocol based on fatigue life-extension. Significant crack arrest and life-extension resulted when the in situ healing rate was faster than the crack-growth rate. In loading cases where the crack grew too rapidly (maximum applied stress intensity factor is a significant percentage of the mode-I fracture toughness value), carefully timed rest periods were used to prolong fatigue life. At lower values of applied stress intensity factor, crack growth was arrested completely. The self-healing material system demonstrated great potential for extending component life under fatigue loading, with the degree of life-extension dependent on a number of interrelated variables such as stress amplitude, frequency, in situ healing rate, and rest periods.

Fatigue life-extension from in situ self-healing was achieved by a combination of crack-tip shielding mechanisms. First, viscous flow of the healing agent in the crack plane retarded crack growth. Second, polymerization of the healing agent provided a short term adhesive effect and a long term crack closure effect, which prevented unloading of the crack tip. Successful healing

resulted in reduced crack length and retardation of additional crack growth. These shielding mechanisms also contributed to increasing the threshold ΔK_{th}, effectively increasing the amplitude of applied ΔK_I that the material can be subjected to without crack propagation. Above the threshold ΔK_{th}, the self-healing shielding mechanisms led to temporary arrest of fatigue crack growth and significantly extended fatigue life.

The dominant shielding mechanism observed for in situ self-healing was crack closure induced by the formation of a polyDCPD wedge at the crack tip. Similar to the case of artificial crack closure achieved by manual injection, the profile of the load–displacement curves progresses from linear to bi-modal as the crack propagates through the healed region.

The success of crack closure was strongly dependent on how efficiently the crack tip was shielded from the applied cyclic loads. A polymer wedge formed at or above K_{max} provided maximum shielding, approaching a stress free crack tip. Conversely a polymer wedge polymerized at zero-load provided minimal crack tip shielding. Because the polymer wedge formed under continuous cyclic loading was created between moving boundaries, the wedge structure was irregular and resulted in shielding efficiency between the two extremes.

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